

INTERNATIONAL STANDARD

ISO 14218

First edition
1998-05-15

Aerospace — Airframe ball bearings, single-row, self-aligning, sealed, heavy duty — Inch series

*Aéronautique et espace — Roulements pour structures d'aéronefs:
roulements, à rotule, sur une rangée de billes, avec joints d'étanchéité,
série lourde — Série en inches*



Reference number
ISO 14218:1998(E)

Foreword

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Draft International Standards adopted by the technical committees are circulated to the member bodies for voting. Publication as an International Standard requires approval by at least 75 % of the member bodies casting a vote.

International Standard ISO 14218 was prepared by Technical Committee ISO/TC 20, *Aircraft and space vehicles*, Subcommittee SC 15, *Airframe bearings*.

Annex A of this International Standard is for information only.

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Introduction

At the time this International Standard was developed, the Imperial unit sizes of airframe ball bearings were dominant in world application. The basis for this International Standard is the Imperial units provided in annex A. For new applications, the use of metric series airframe ball bearings is encouraged.

Aerospace — Airframe ball bearings, single-row, self-aligning, sealed, heavy duty — Inch series

1 Scope

This International Standard specifies the characteristics, boundary dimensions, tolerances, internal clearances and permissible loads of inch series, single-row, self-aligning, sealed, heavy duty ball bearings used in airframe applications. These bearings are full complement (without cage) with a single row of balls. These bearings are designed to withstand only slow rotations and oscillations under radial load and are intended for use between fixed and moving parts of an aircraft structure and their control surfaces.

The airframe ball bearings covered by this International Standard are designed to operate in the temperature range of – 54 °C to + 150 °C.

2 Normative references

The following standards contain provisions which, through reference in this text, constitute provisions of this International Standard. At the time of publication, the editions indicated were valid. All standards are subject to revision, and parties to agreements based on this International Standard are encouraged to investigate the possibility of applying the most recent editions of the standards indicated below. Members of IEC and ISO maintain registers of currently valid International Standards.

ISO 683-17:—¹⁾, *Heat-treated steels, alloy steels and free-cutting steels — Part 17: Ball and roller bearing steels.*

ISO 1132:1980, *Rolling bearings — Tolerances — Definitions.*

ISO 2082:1986, *Metallic coatings — Electroplated coatings of cadmium on iron or steel.*

ISO 4520:1981, *Chromate conversion coatings on electroplated zinc and cadmium coatings.*

ISO 5593:1997, *Rolling bearings — Vocabulary.*

ISO 8075:1985, *Aerospace — Surface treatment of hardenable stainless steel parts.*

ISO 14190:1998, *Aerospace — Airframe rolling bearings: ball and spherical roller bearings — Technical specification.*

AMS 2417E:1993, *Plating, zinc-nickel alloy.*²⁾

¹⁾ To be published. (Revision of ISO 683-17:1976)

²⁾ Available from: SAE International
400 Commonwealth Drive
Warrendale, PA 15096-0001
USA

3 Definitions

For the purposes of this International Standard, the definitions given in ISO 5593 apply.

4 Symbols

4.1 For the purposes of this International Standard, the symbols given in ISO 1132 apply. The symbols (except those for tolerances) shown in figure 1 and the values given in tables 1 and A.1 denote nominal dimensions unless specified otherwise.

4.2 Additional symbols for bearings covered by this International Standard are the following.

C_s permissible static radial load

$F_{a \max}$ permissible static axial load

5 Required characteristics

5.1 Dimensions — Tolerances — Internal clearances — Loads

Values are given in table 1. Where plating is specified, tolerances are applicable after plating.

Configuration is given in figure 1. The configuration for and installation process of seals are at the manufacturer's option.

5.2 Surface roughness

Inner and outer ring raceways and rolling elements shall have a surface roughness of $R_a = 0,2 \mu\text{m max}$.

Bore, side faces and cylindrical outer surface shall have a surface roughness of $R_a = 0,8 \mu\text{m max}$ before surface treatment.

6 Materials

6.1 Low alloy through hardening bearing steel (code letter F, D, M and Z)

Rings: bearing steel — ISO 683-17, type 1, surface hardness 59 HRC to 64 HRC (710 HV to 840 HV).

Balls: bearing steel — ISO 683-17, type 1, heat treated to 59 HRC to 64 HRC (710 HV to 840 HV).

6.2 Corrosion-resistant stainless steel (code letters C and H)

Rings: bearing steel — ISO 683-17, type 21, hardness $\geq 58 \text{ HRC}$ ($\geq 670 \text{ HV}$).

Balls: bearing steel — ISO 683-17, type 21, hardness $\geq 58 \text{ HRC}$ ($\geq 670 \text{ HV}$).

6.3 General

Seals: polytetrafluoroethylene (PTFE) or polytetrafluoroethylene (PTFE) sheet, glass-fabric reinforced.

Seal retainers: corrosion-resistant steel.

7 Surface treatment

7.1 All elements manufactured from corrosion-resistant steel which are not cadmium plated or zinc-nickel plated (code letter C) shall be passivated in accordance with ISO 8075 or cleaned by a mechanical method.

7.2 If made of corrosion-resistant steel, cadmium plating (code letter H) shall be subject to agreement between the customer and the manufacturer.

7.3 Where cadmium plating is specified (code letters D, M and H), it shall be in accordance with ISO 2082. The external surfaces of bearing rings, except the bore of the inner ring, shall be cadmium plated. Plating on the internal surfaces shall be at the manufacturers option, except the raceways shall not be plated. The thickness of the plating shall not be less than 7 µm and not more than 15 µm, except plating on the chamfers of the bore may vary from the specified thickness. The bearing shall be embrittlement-relieved within 4 h of plating by heat treatment at 140 °C ± 10 °C for a minimum of 8 h followed by chromate treatment in accordance with ISO 4520 (code letters D and H only).

7.4 Where zinc-nickel plating is specified (code letter Z), it shall be in accordance with AMS 2417E, type 2. The surface coverage of the plating shall be as specified in 7.3. The thickness of the plating shall not be less than 7 µm or more than 15 µm.

8 Lubrication

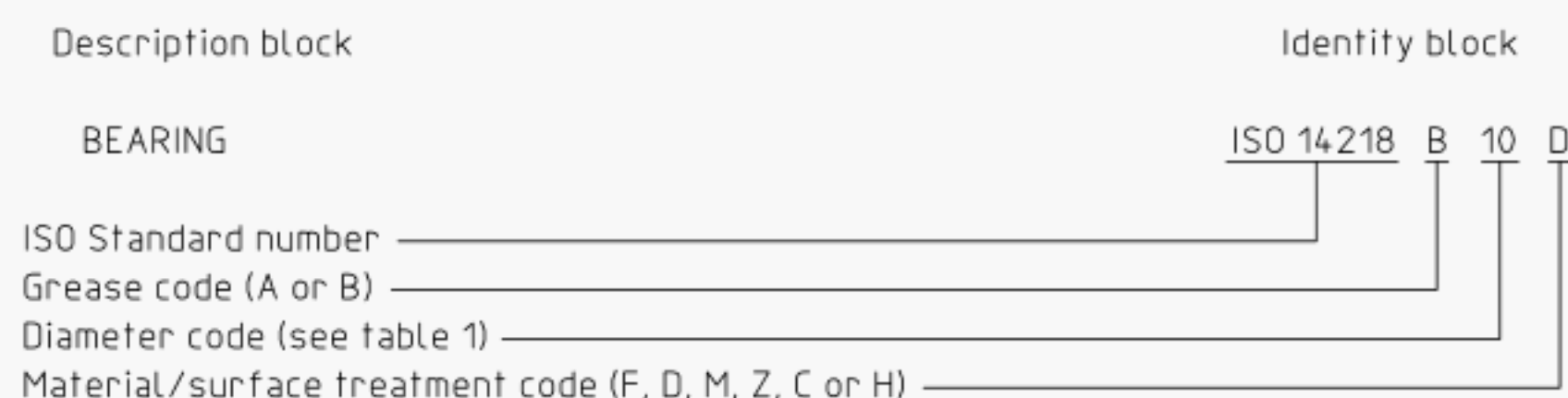
The bearing shall be prelubricated with either grease A or B, as specified by the customer.

NOTE — Descriptions of grease A and grease B are given in ISO 14190:1998, annex H.

9 Designation

Bearings in accordance by this International Standard shall be designated only as shown in the following example.

EXAMPLE



where the following codes are applied:

— greases:

A = ester type grease;

B = synthetic hydrocarbon type grease;

— materials/surface treatments:

F = material: low alloy bearing steel;

surface treatment: none;

D = material: low alloy bearing steel;

surface treatment: cadmium plated with chromate treatment;

M = material: low alloy bearing steel;

surface treatment: cadmium plated without chromate treatment;

- Z = material: low alloy bearing steel;
 surface treatment: zinc-nickel plated;
- C = material: corrosion-resistant stainless steel;
 surface treatment: none;
- H = material: corrosion-resistant stainless steel;
 surface treatment: cadmium plated with chromate treatment.

10 Identification marking

In addition to the manufacturer's name or trademark, each bearing shall be permanently and legibly marked, using the identity block as defined in clause 9. Marking position and method are at the manufacturer's option.

11 Technical specification

Airframe ball bearings supplied to this International Standard shall conform to the requirements of ISO 14190.

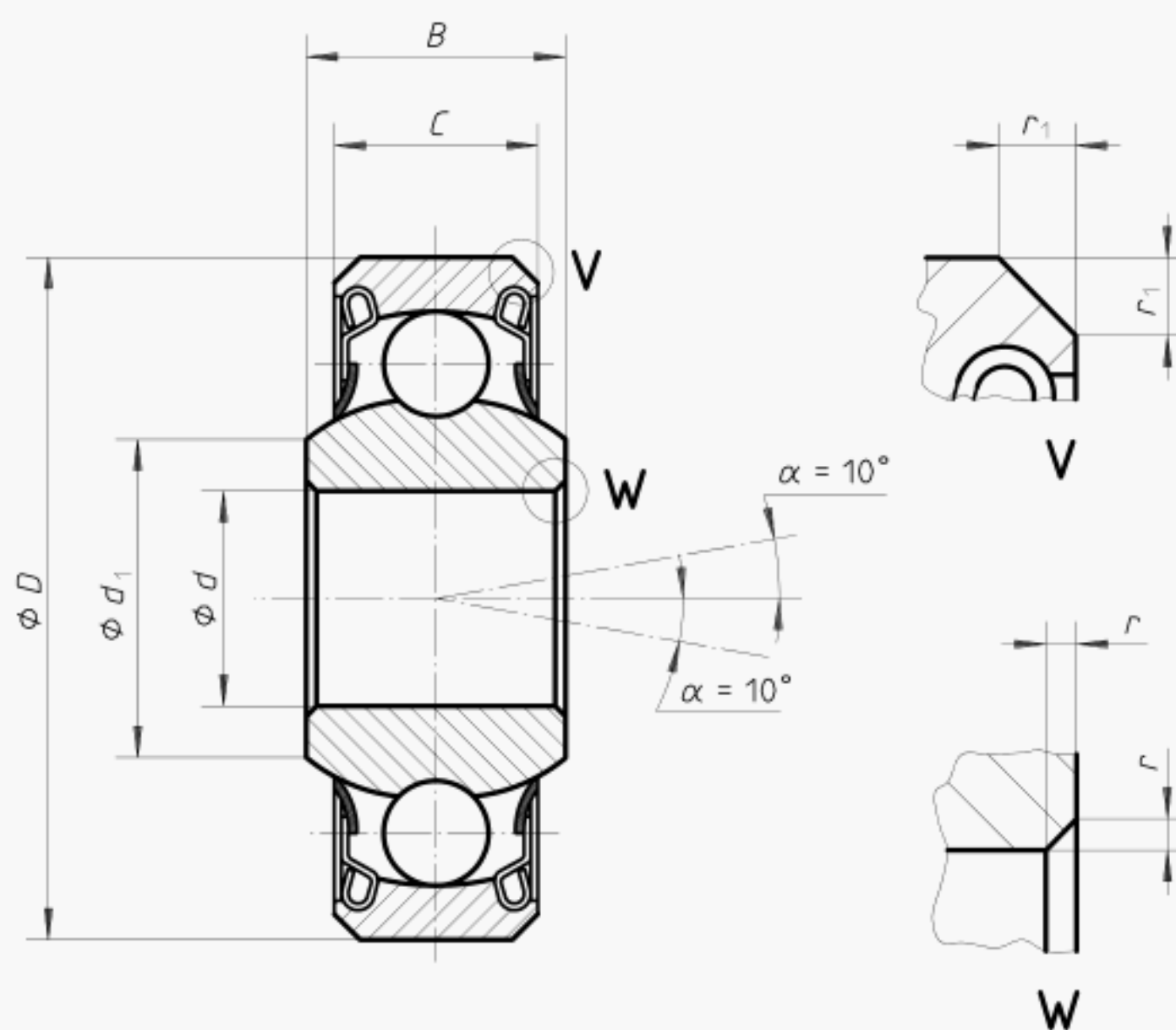


Figure 1

Table 1

Dimensions in millimetres,
tolerances in micrometres

Diameter code	d + 5 − 18	D + 13 − 25	B 0 − 127	C 0 − 127	Δ_{dmp}	Δ_{Dmp}	V_{Bs} max.	V_{Cs} max.	d_1 ≈	r	r_1	Mass g ≈
03	4,826	19,746	7,54	6,86	0 − 13	0 − 13	20	20	7,62	0,56 to 0,94	0,13 to 0,51	14
04	6,35	22,896	12,29	8,51					10,16	0,81 to 1,19		0,38 to 0,76
05	7,938	31,75	14,17	9,52					14,48		45	
06	9,525	36,512	15,75	11,91					15,49			
08	12,7	42,862		12,7					20,32	1,12 to 1,5	104	
10	15,875	49,212	20,65	15,88					23,37			

Diameter code	Internal clearances		Radial runout tolerances		Starting torque	Permissible static radial load	Permissible static axial load
	μm		max. μm		max.	C_s	$F_{a \text{ max}}$
	Radial G_r	Axial $G_{a \text{ max.}}$	K_{ia}	K_{ea}	mN·m	kN	kN
03	0 to 25	762	25	41	14,1	4	0,89
04		838				6,27	1,33
05		965			21,2	9,74	
06		1 041			28,2	13,26	1,78
08		1 118			35,3	16,32	2,22
10		1 194			42,4	23,66	2,67

Annex A
(informative)
Imperial unit information

Table A.1 gives the Imperial (non-SI) unit information that is the basis for the normative portion of this International Standard.

Table A.1

Dimensions in inches,
tolerances in 0,000 1 inches

Diameter code	d + 2 − 7	D + 5 − 10	B 0 − 50	C 0 − 50	Δ_{dmp}	Δ_{Dmp}	V_{Bs} max.	V_{Cs} max.	d_1 ≈	r	r_1	Mass lb ≈
03	0,19	0,777 4	0,297	0,27	0 − 5	0 − 5	8	8	0,3	0,022 to 0,037	0,005 to 0,02	0,03
04	0,25	0,901 4	0,484	0,335					0,4	0,032 to 0,047		0,015 to 0,03
05	0,312 5	1,25	0,558	0,375					0,57		0,1	
06	0,375	1,437 5	0,62	0,469					0,61			
08	0,5	1,687 5		0,5					0,8		0,044 to 0,059	
10	0,625	1,937 5	0,813	0,625					0,92	0,37		

Diameter code	Internal clearances		Radial runout tolerances		Starting torque	Permissible static radial load	Permissible static axial load
	0,000 1 in		max. 0,000 1 in		max.	C_s	$F_{a \max}$
	Radial G_r	Axial G_a max.	K_{ia}	K_{ea}	in·oz	lbf	lbf
03	0 to 10	300	10	16	2	900	200
04		330				1 410	300
05		380			3	2 190	
06		410			4	2 980	400
08		440			5	3 670	500
10		470			6	5 320	600

ICS 49.035

Descriptors: aircraft industry, rolling bearings, airframe bearings, ball bearings, single-row bearings, self-aligning bearings, specifications, materials specifications, characteristics, load capacity, dimensions, overall dimensions, lubrication, designation, marking, imperial system, heavy duty series.

Price based on 6 pages
